

AS365

Emergency Access and Rescue from Helicopter



IMPORTANT NOTE

This Ground Rescue Booklet provided by Airbus provides general and safety information concerning the AS365 series helicopter. This document shall only be considered as a support for users to prepare their own documentation. It will not be systematically updated according to aircraft modification process.

Depending on the country and the configuration of the helicopter, systems may differ in their location.

This information booklet is provided free of charge by Airbus.

Wide-spread dissemination to firefighters and rescue teams around the world is strongly encouraged. Copies can be downloaded from the Airbus Helicopters web site.

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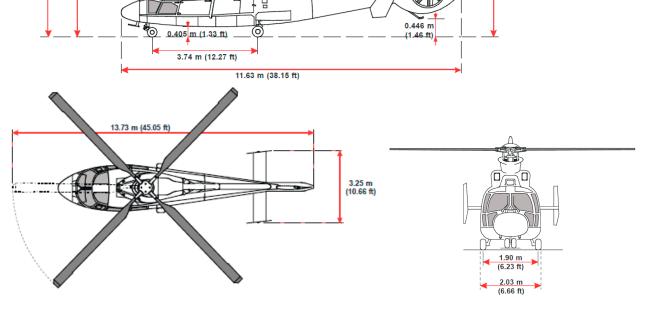


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1	GENERAL INFORMATION
	MAXIMUM GROSS WEIGHT up to 9480 Lbs. / 4300 Kg
	EMPTY WEIGHT up to 5357 Lbs. / 2430 Kg
	OCCUPANCY
	- CrewOne pilot or two pilots
	 Commercial: Standard Version Utility Version 1 Pilot + 9 passengers Utility Version 1 Pilot + 13 Passengers Medevac/EMS transport: 1 pilot + 1 copilot + 1 medical stretcher + medical attendant + 2 passengers 1 pilot + 1 medical stretcher + medical attendant + 2 passengers 1 pilot + 1 medical stretcher + medical attendant + 2 passengers
	DIMENSIONS
	Fuselage length:11.63 mFuselage width:2.03 mRotor diameter:11.94 mFenestron height:up to 3.97 m
	11.94 m (39.17 ft) $4 m (39.17 ft)$ $3.80 m (12.49 ft)$ $3.80 m (12.49 ft)$





LIFTING AIRCRAFT

Lift the aircraft with the specific STARFLEX sling. - Ballast the aircraft center of gravity

- Guide the aircraft ropes





TOWING AIRCRAFT

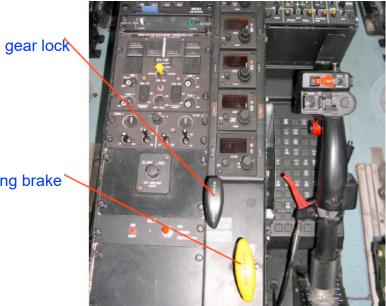


CAUTION: NOSE GEAR LOCK AND BRAKE HANDLES MUST BE DOWN BEFORE TOWING AIRCRAFT.

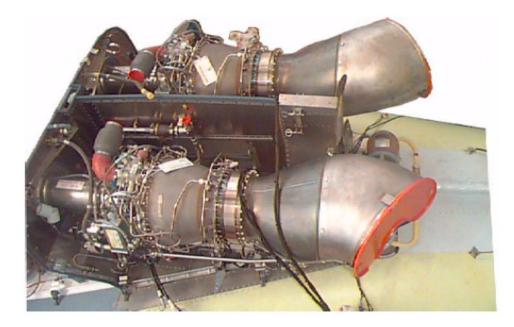


Nose gear lock

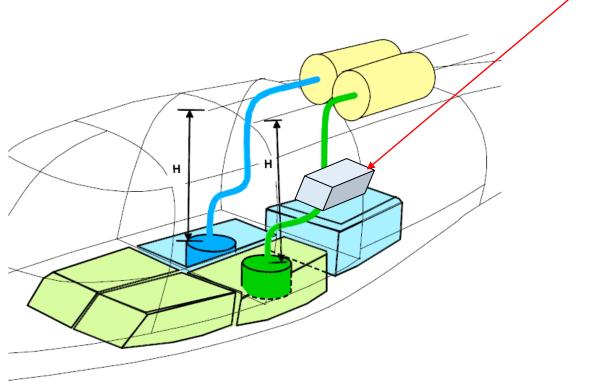
Parking brake







FUEL CAPACITY	Up to 1158 liters
	LH group (green): 573 liters
	RH group (blue): 585 liters



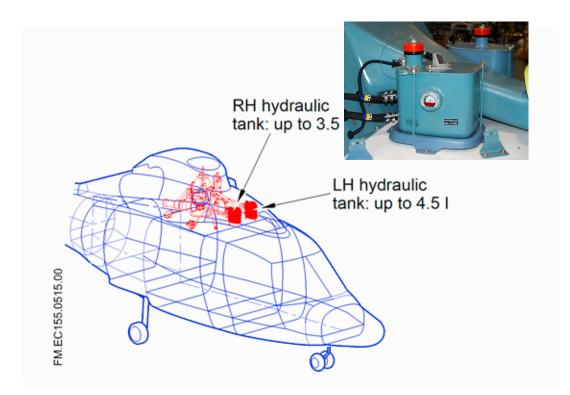


OIL CAPACITY

Engine oil	6.2 liters
Main Transmission	9.5 liters
Tail gear box	0.5 liters

HYDRAULIC FLUID CAPACITY

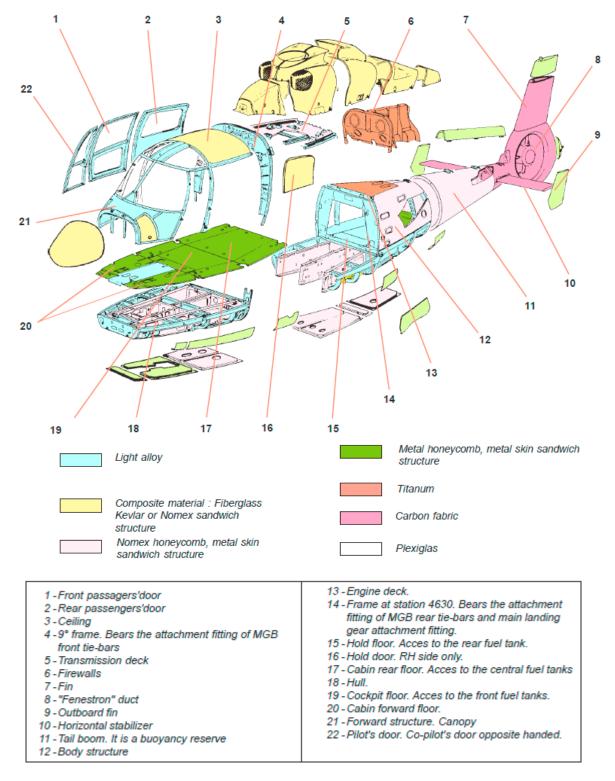
Left Hydraulic System	4.5 liters
Right Hydraulic System	3.5 liters



(RH/LH = Right Hand side/Left Hand side)



MATERIALS



The wide use of nonmetallic materials should be noted.



2 SAFETY INFORMATION - OUTSIDE THE AIRCRAFT

AIRCRAFT MAY BE CHARGED WITH STATIC ELECTRICITY. USE GLOVES AND IF POSSIBLE DISCHARGE THE AIRCRAFT BY ESTABLISHING AN ELECTRICAL GROUNDING.

DANGER AREA WITH ROTOR TURNING



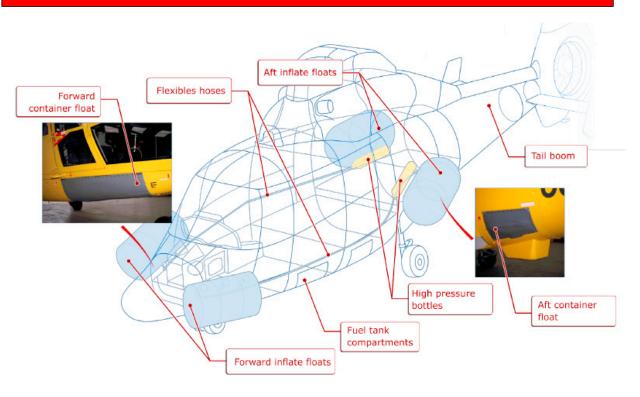




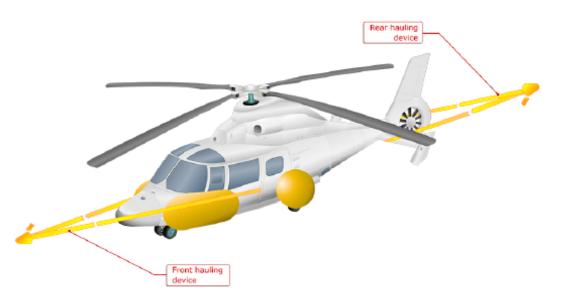


EMERGENCY FLOATATION GEAR

FLOATS MAY INFLATE SUDDENLY. THE PRESSURE CYLINDERS ARE EACH FILLED WITH HELIUM.



WATER TOWING instructions in case of ditching





ΡΙΤΟΤ

PITOT IS HEATED IN FLIGHT AND CAN CAUSE BURNS.



OR (according to version)





FIRE FIGHTING RECOMMENDATIONS

GENERAL

- 1) GROUND STAFF MUST BE IN CONTACT (RADIO / VISUAL SIGNS) WITH THE AIRCREW IN ORDER TO COORDINATE AND SECURE THE INTERVENTION.
- 2) GROUND STAFF MUST WEAR ADEQUATE PROTECTIVE EQUIPMENT.

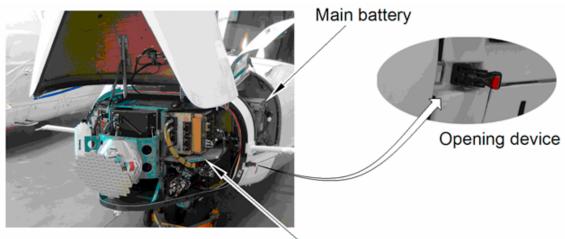
FIRE AROUND THE AIRCRAFT

If possible, wait for the rotor to come a complete stop.

FUEL LEAKAGE ALONG THE AIRCRAFT STRUCTURE AND/OR PRESENCE OF FIRE SPILL ON GROUND MUST BE FOUGHT FIRST WITH FOAM.

- Cool external adjacent structures with foam or water spray.

FIRE IN THE FRONT COMPARTMENT



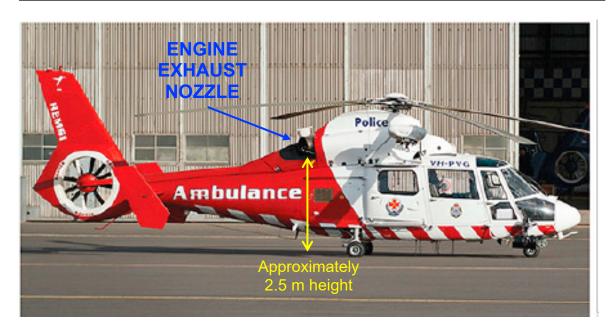
Avionics bay

- Slowly open the front compartment (radome) cowling to avoid a sudden supply of oxygen and a flash-over.
- Saturate the compartment with the extinguishing agent (gaseous extinguisher recommended).



FIRE IN THE ENGINE COMPARTMENT

- 1) WAIT FOR ENGINES AND ROTOR FULL STOP.
- 2) THE TEMPERATURE OF THE ENGINE EXHAUST NOZZLE COULD BE VERY HOT (UP TO 600°C).





- Spray the extinguishing agent (gaseous extinguisher recommended) between engine exhaust and engine nozzle.
- Proceed with circular movements until saturation occurs.

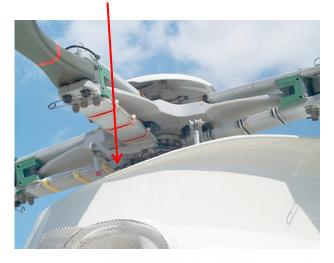


FIRE IN THE MAIN GEAR BOX (MGB) COMPARTMENT

WAIT FOR ENGINES AND ROTOR FULL STOP.

Possible access for extinguishing

MGB compartment







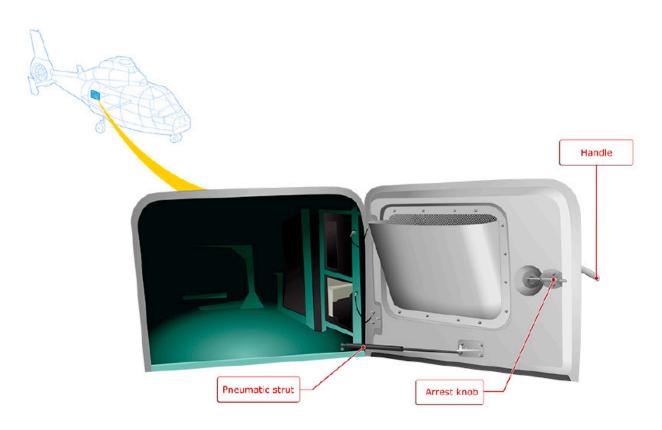
- Spray the extinguishing agent through the easiest available way (gaseous extinguisher recommended) for saturating the MGB compartment. Do not try to open the cowlings. In case of severe flash-over, use foam.

Revision 0



FIRE IN THE LUGGAGE HOLD

REMINDER : DO NOT TRY TO OPEN THE LUGGAGE HOLD WITH THE ROTORS SPINNING.



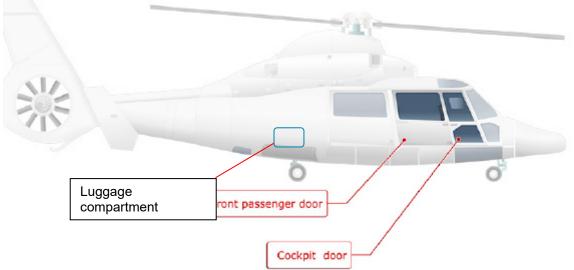
- Spray the luggage hold with the extinguishing agent (gaseous extinguisher recommended).



EMERGENCY ACCESS

REMINDER : DO NOT TRY TO OPEN THE LUGGAGE HOLD WITH THE ROTORS SPINNING.





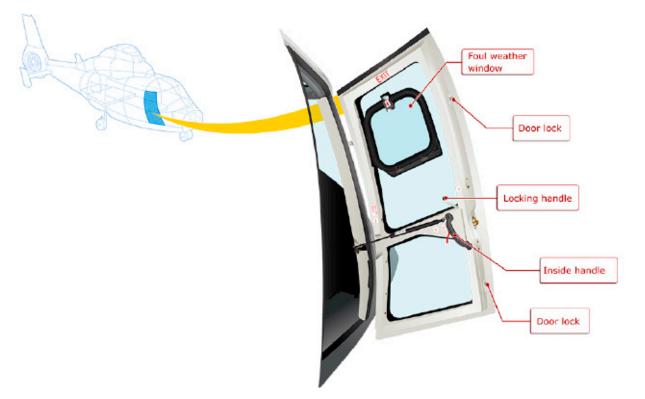
The aircraft has three doors on each side:

- Cockpit door,
- Front passenger door,
- Rear flap or sliding passenger door.

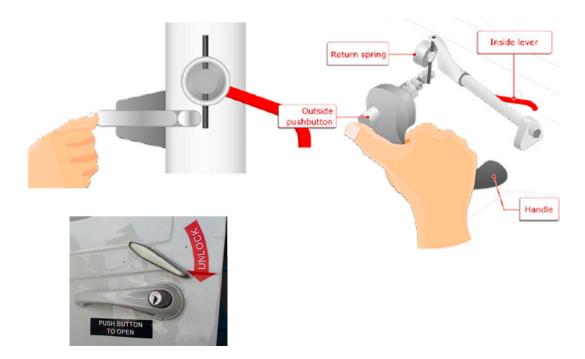


COCKPIT DOORS

Opening Cockpit Doors



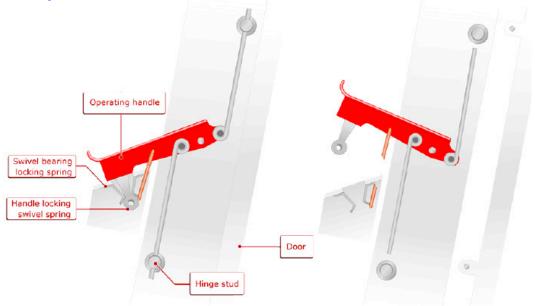
Front doors can be opened by actuating the handles from the inside or the pushbutton outside.



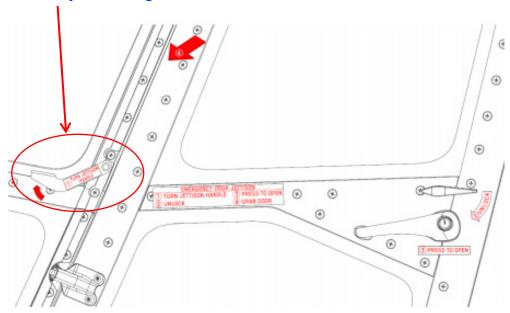


Jettisoning Cockpit Doors

Pilot and copilot doors can be jettisoned by actuating the Jettisoning lever from inside the aircraft. It causes the door to fall away.



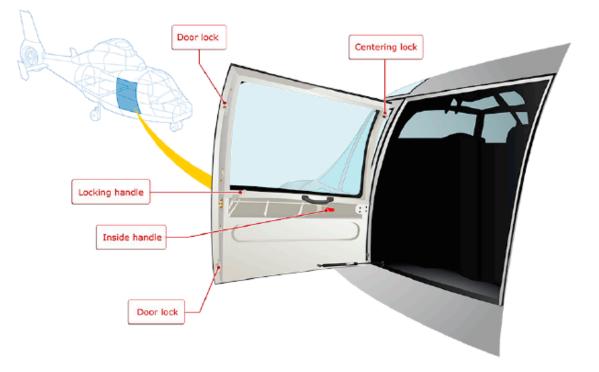
Aircraft equipped with emergency floatation system may have the possibility to jettison the doors from outside too:



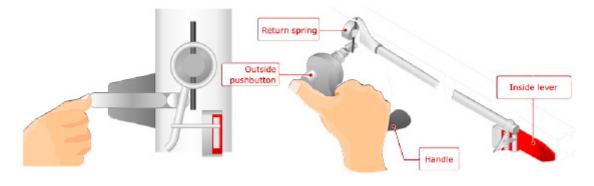
External jettisoning handle



FORWARD PASSENGER DOORS



Forward Passenger Door Locking System

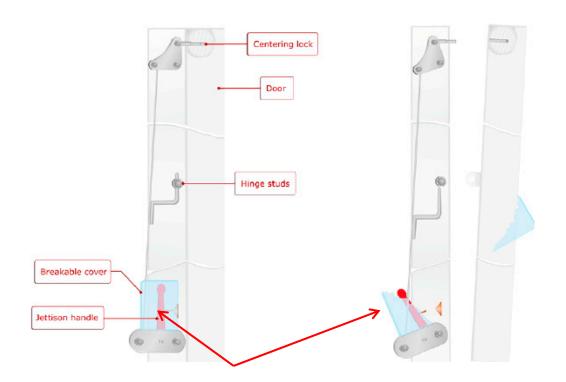






Jettisoning Forward Passenger Doors

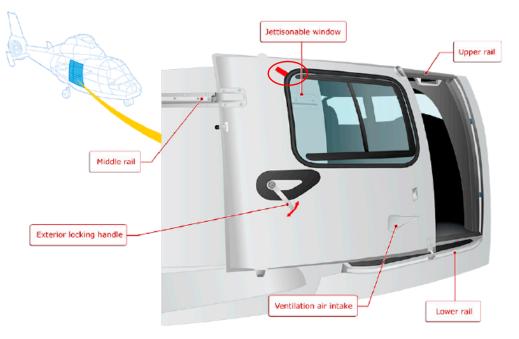
Passenger doors can only be jettisoned by actuating the Jettisoning handle from inside the aircraft. It causes the door to fall away.



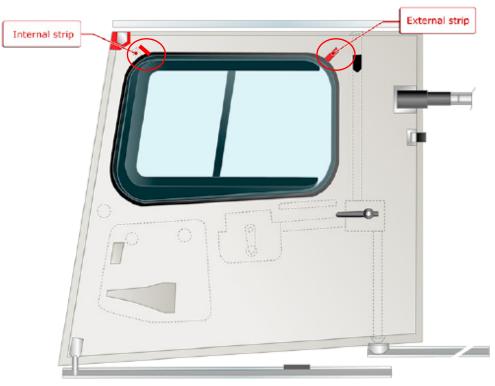


SLIDING DOORS

Opening Sliding Doors



Sliding Door Window Jettison



Sliding door window is Jettisonable, from inside or outside by pushing out / pulling out strongly after removing the **red** jettison retaining strips.



LUGGAGE COMPARTMENT DOOR

The luggage hold is accessible on the right side of the helicopter.



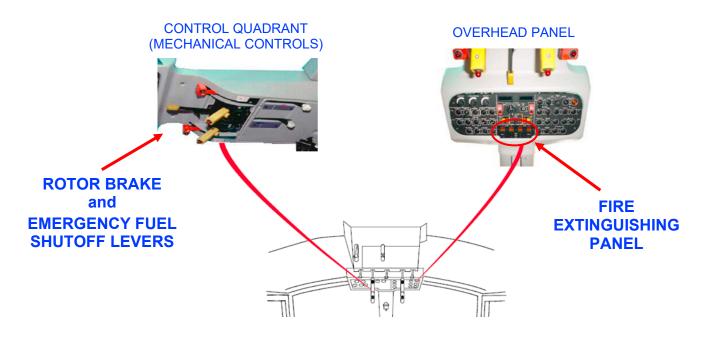
Compensation rod

Lock



3 SAFETY INFORMATION - INSIDE THE AIRCRAFT

COCKPIT LAYOUT



ELECTRICAL CONTROL PANEL : EMERGENCY CUTOFF BATTERY SWITCHES

PARKING BRAKE HANDLE: - Down : brakes released - Up : brakes set



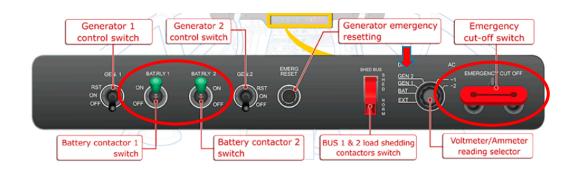




THE FOLLOWING PROCEDURES ARE TO BE USED IN CASE OF EMERGENCY ON GROUND ONLY, IF PILOTS ARE INCAPACITED.

ELECTRICAL SHUTDOWN

- Both (GREEN) Battery switchesOFF
- Emergency cut-off switchOFF



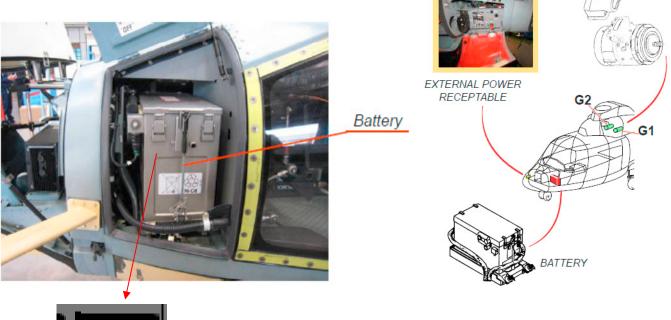
OR (according to version)





BATTERY

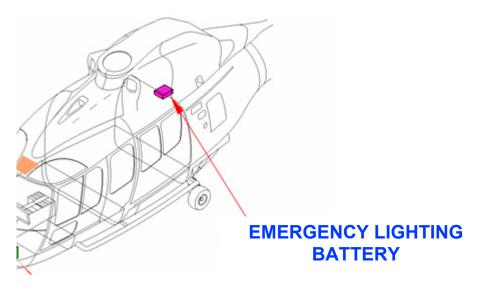
Main battery is located between the nose and the cockpit of the aircraft.





CAUTION

DISCONNECT BATTERY ONLY WHEN THE ENGINES ARE SWITCHED OFF AND ROTOR IS STOPPED.





ENGINE SHUTDOWN PROCEDURE

AS365N3/N3+

Engines with electronic management system computer

- Both engine control switches<mark>OFF</mark>



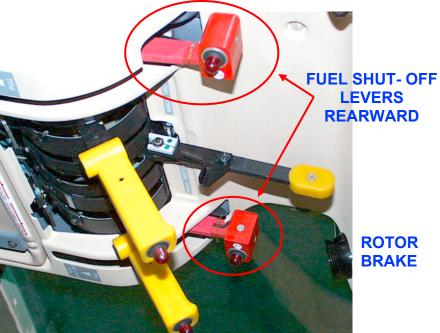
- Both emergency fuel shut-off levers Rearward











OFF / IDLE / FLT **Engine control switches**



AS365N/N1/N2

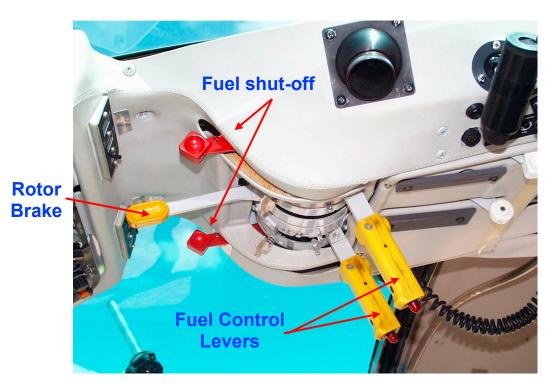
Engine with mechanical fuel management system.

- Move the two fuel control levers fully aft to stop engines, before applying the rotor brake control lever.

ROTOR BRAKING

ENGINES MUST BE STOPPED BEFORE APPLYING ROTOR BRAKE.

- Move the center yellow rotor brake lever rearwards to stop the rotor.



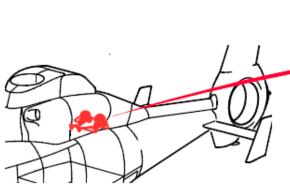
Less than 170 RPM before applying rotor brake.





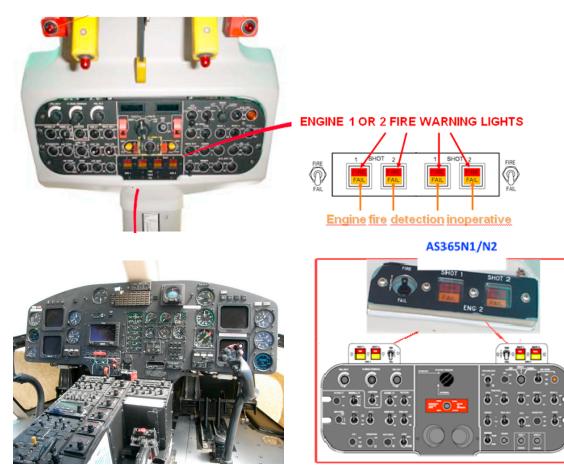
ENGINE FIRE DETECTION AND EXTINGUISHING SYSTEM

IN CASE OF ENGINE FIRE DETECTION, APPLY THE ENGINE SHUTDOWN PROCEDURE FIRST.





The system consists of detection and extinguishing circuits with two Freon 13B1 fire extinguishers.



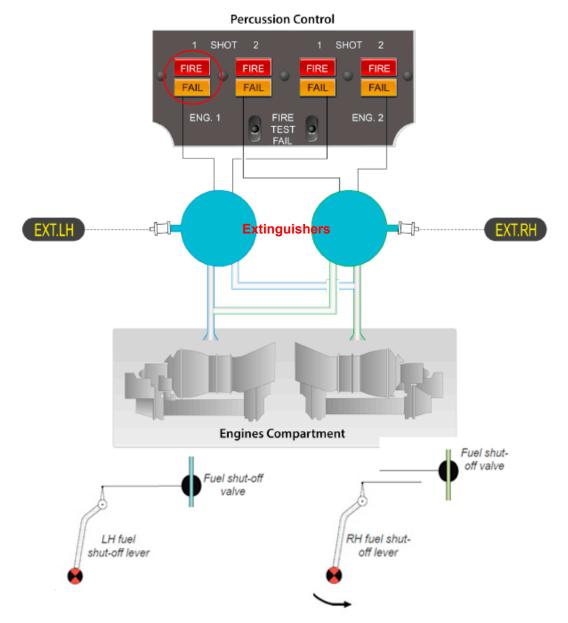


PROCEDURE IN CASE OF ENGINE FIRE DETECTION

- 1) Fuel Shut-off...... closed
- 2) Rotor brakeapplied (NR below 170 rpm)
- 3) Fight fire from outside if possible.

If the fire cannot be fought from the outside:

- ON THE AFFECTED ENGINE PRESS SHOT 1
- THEN, IF FIRE RED LIGHT REMAINS ON,
 - AFTER A 10s DELAY..... PRESS SHOT 2





SAFETY BELTS



To release the safety belt, turn the center lock until each belt is free.

